

Adaptive Controllers and Robustness Analysis for Curve Tracking with Unknown Control Gains

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Abstract—We study adaptive control and parameter identification for robotic curve tracking under unknown control gains. We build adaptive controllers that identify the unknown control gains and stabilize equilibria corresponding to a fixed constant distance to the curve and zero bearing. Our strict Lyapunov function method allows us to prove robust performance under actuator errors in terms of integral input-to-state stability under a bound on the disturbance that maintains forward invariance of a class of invariant hexagons. This extends existing curve tracking results to allow controller uncertainty and parameter identification. We demonstrate our work in simulations.

Key Words—Curve tracking, robotics, robustness

I. INTRODUCTION

Curve tracking methods have proven to be effective for navigating mobile robots in complex environments [5], [15], [17]. For wheeled mobile robots, feedback control laws have been developed that achieve autonomous tracking of obstacle boundaries or a desired smooth path [10]. There were several recent improvements for curve tracking control [11]. These include extensions to the three dimensional case [3] and cooperative control for ocean sensing [16], [18].

Evidence of robustness in real life applications has been reported for farming, obstacle avoidance in corridors, and ocean sampling [4], [16], [19]. Our previous work [7] theoretically justified the robustness by showing that the feedback control laws for two-dimensional curve tracking from [17] provide input-to-state stability (ISS) properties with respect to additive uncertainty on the controller. We proved robust forward invariance of a nested sequence of hexagons $H_1 \subseteq H_2 \subseteq H_3 \subseteq \dots$ that fill the state space. This produced a sequence of positive constants $\{\delta_i^*\}$ such that the following holds for all i : all trajectories of the curve tracking dynamics that start in H_i and have uncertainties bounded by δ_i^* remain in H_i for all positive times. Then we proved ISS of the curve tracking dynamics on each hexagon H_i , with respect to disturbances bounded by δ_i^* . This theoretical understanding is important for giving predictable tolerance and safety bounds for the curve tracking control laws. See [12]–[14] for background on ISS.

This paper continues our search for robust controller designs for curve tracking dynamics that respect the constraints

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that prevail in real time experimental implementations. We are motivated by our recent deployment of a fleet of marine robots at Grand Isle State Park, Louisiana which surveyed residual crude oil components from the Deepwater Horizon oil spill disaster [2]. In this application, we implemented curve tracking behaviors on two different marine robots with unknown actuator parameters. We implemented the curve tracking control laws from [7], [17] by testing different control gains, and the experimental results will be reported in other publications.

However, the control laws in [7], [17] would be more applicable to real systems if they could be made adaptive to unknown actuation parameters. In this paper, we present an adaptive controller design that can be implemented on the marine robots. We also prove that the adaptive control laws are integral ISS (iISS) with respect to actuation errors. Our adaptive curve tracking result is based on an extension of the barrier Lyapunov function approach from [9] to systems with state constraints, combined with an adaptive version of the robust forward invariant hexagon argument from [7], and includes parameter identification. It differs from the usual adaptive control problem of stabilizing an equilibrium point while keeping the estimate of the unknown parameter bounded, which is more common in the literature. For complete proofs of all results to follow, see [8].

II. NOTATION AND DEFINITIONS

A continuous function $\gamma : [0, +\infty) \rightarrow [0, +\infty)$ is of class \mathcal{H} (written $\gamma \in \mathcal{H}$) provided it is strictly increasing and $\gamma(0) = 0$; if, in addition, $\gamma(r) \rightarrow +\infty$ as $r \rightarrow +\infty$, then it is of class \mathcal{H}_∞ (written $\gamma \in \mathcal{H}_\infty$). A continuous function $\beta : [0, +\infty) \times [0, +\infty) \rightarrow [0, +\infty)$ is of class \mathcal{HL} (written $\beta \in \mathcal{HL}$) provided (i) for each fixed $s \geq 0$, the function $\beta(\cdot, s) \in \mathcal{H}$ and (ii) for each fixed $r \geq 0$, the function $\beta(r, \cdot)$ is non-increasing and $\beta(r, s) \rightarrow 0$ as $s \rightarrow +\infty$. Let \mathcal{U} be any subset of a Euclidean space such that $0 \in \mathcal{U}$, and let $\mathcal{M}_{\mathcal{U}}$ denote the set of all measurable locally essentially bounded functions $\delta : [0, +\infty) \rightarrow \mathcal{U}$. Let $|f|_{\mathcal{S}}$ denote the essential supremum of the restriction of any function f to any subset \mathcal{S} of its domain.

Consider any subset \mathcal{G} of a Euclidean space and any point $\mathcal{E} \in \mathcal{G}$. A continuous function $\mathcal{V} : \mathcal{G} \rightarrow [0, +\infty)$ is *positive semi-definite* with respect to \mathcal{E} provided $\mathcal{V}(\mathcal{E}) = 0$; if, in addition, $\mathcal{V}(q) > 0$ for all $q \in \mathcal{G} \setminus \{\mathcal{E}\}$, then \mathcal{V} is *positive definite* with respect to \mathcal{E} . A function \mathcal{V} is *negative semi-definite* (resp., *definite*) with respect to \mathcal{E} provided $-\mathcal{V}$ is positive semi-definite (resp., definite) with respect to \mathcal{E} . Let $|p|_{\mathcal{E}} = |p - \mathcal{E}|$ denote the distance between any point $p \in \mathcal{G}$

and the point \mathcal{E} in the usual Euclidean metric. A function $\mathcal{V} : \mathcal{G} \rightarrow [0, +\infty)$ is a *modulus* with respect to $(\mathcal{E}, \mathcal{G})$ provided it is positive definite with respect to \mathcal{E} and *radially unbounded* in the following sense: For each constant $M > 0$, there is a constant $\delta_M > 0$ (depending on M) such that $\mathcal{V}(x) \geq M$ for all $x \in \mathcal{G}$ that satisfy either $\text{distance}(x, \text{boundary}(\mathcal{G})) \leq \delta_M$ or $|x|_{\mathcal{E}} \geq 1/\delta_M$. For any subset $\mathcal{S} \subseteq \mathbb{R}^n$ and any point $\bar{p} \in \mathbb{R}^n$, we set $\mathcal{S} - \bar{p} = \{q - \bar{p} : q \in \mathcal{S}\}$.

Consider a forward complete system $\dot{x} = \mathcal{F}(x, \delta)$ with state space \mathcal{G} and disturbances $\delta \in \mathcal{M}_{\mathcal{U}}$, where $\mathcal{F} : \mathcal{G} \times \mathcal{U} \rightarrow \mathcal{G}$ satisfies the standard existence and uniqueness of solutions properties for all initial states $x_0 \in \mathcal{G}$ and all disturbances $\delta \in \mathcal{M}_{\mathcal{U}}$ [6], and $\mathcal{F}(\mathcal{E}, 0) = 0$. Let $\mathcal{S} \subseteq \mathcal{G}$ be any neighborhood of \mathcal{E} . We say that the system is *integral input-to-state stable (iISS)* with respect to $(\mathcal{U}, \mathcal{E}, \mathcal{S})$ provided there are functions $\beta \in \mathcal{KL}$ and $\gamma_i \in \mathcal{K}_{\infty}$ and a modulus Λ with respect to \mathcal{S} such that $\gamma_1(|x(t, x_0, \delta)|_{\mathcal{E}}) \leq \beta(\Lambda(x_0), t) + \int_0^t \gamma_2(|\delta(r)|) dr$ for all $t \geq 0$, all solutions $x(t, x_0, \delta)$ of the system with initial states $x_0 \in \mathcal{S}$, and all $\delta \in \mathcal{M}_{\mathcal{U}}$. This agrees with the usual iISS condition from [1], [13] when $\mathcal{G} = \mathcal{S} = \mathbb{R}^n$ and $\mathcal{E} = 0$. The special case of iISS where \mathcal{F} only depends on x and the integral term in the iISS estimate is not present is *global asymptotic stability (GAS)* with respect to $(\mathcal{E}, \mathcal{S})$. In that case, a *nonstrict* (resp., *strict*) *Lyapunov function* for the system with respect to $(\mathcal{E}, \mathcal{G})$ is any modulus \mathcal{V} with respect to $(\mathcal{E}, \mathcal{G})$ that is C^1 on some open set containing \mathcal{G} , and that is such that the function $\dot{\mathcal{V}} : \mathcal{G} \rightarrow \mathbb{R}$ defined by $\dot{\mathcal{V}}(q) = \nabla \mathcal{V}(q) \mathcal{F}(q)$ is negative semi-definite (resp., definite) with respect to \mathcal{E} . A set $\mathcal{S} \subseteq \mathcal{G}$ is *robustly forwardly invariant for the system with disturbances in \mathcal{U}* provided all trajectories of the system starting in \mathcal{S} with disturbances $\delta \in \mathcal{M}_{\mathcal{U}}$ remain in \mathcal{S} for all positive times.

III. PROBLEM FORMULATION

We can show from physics principles that the curve tracking dynamics of marine robots on the surface of water can be simplified to the equations [17]

$$\begin{cases} \dot{\rho} &= -\sin(\phi) \\ \dot{\phi} &= \frac{\kappa \cos(\phi)}{1 + \kappa \rho} - K_2 [u_2 + \delta] \end{cases} \quad (1)$$

where ρ is the relative distance, ϕ is the bearing, κ is the (positive) curvature at the closest point, u_2 is the steering control, the control gain $K_2 > 0$ is constant (and possibly uncertain), the perturbation $\delta \in \mathcal{M}_{\mathbb{R}}$ represents controller uncertainty, and the state space is $\mathcal{X} = (0, +\infty) \times (-\pi/2, \pi/2)$. While (1) is feedback linearizable using the new state variable $z = \sin(\phi)$, the feedback linearized system does not respect the state space and so cannot be used [7].

Assuming that K_2 is known, the work [17] designed a feedback control law to achieve asymptotic stabilization of an equilibrium state corresponding to constant distance ($\rho = \rho_0 > 0$) and zero bearing ($\phi = 0$), which occurs when the robot moves parallel to the curve. We call this case the nonadaptive case.

However, K_2 depends on the vehicle design and the environment, and therefore is unknown in most applications.

In this paper, our goals for an adaptive control law are to have K_2 identified in addition to achieving $\rho \rightarrow \rho_0$ and $\phi \rightarrow 0$ as $t \rightarrow \infty$ from all initial states, when $\delta = 0$. Then, we would like to establish iISS properties for the closed loop dynamics with respect to δ under the adaptive control law. This is more challenging than what has been addressed in our previous works [7], [17] because K_2 is unknown, and because we must simultaneously show integral ISS properties with respect to δ and identify K_2 .

We solve this adaptive stabilization and parameter identification problem using a variant of the barrier nonstrict Lyapunov function and ‘strictification’ methods from [9]. The methods from [9] include parameter identification and integral ISS results with respect to disturbances. However, [9] cannot be applied to (1) directly, since we must ensure that its trajectories $(\rho(t), \phi(t))$ remain in \mathcal{X} . Therefore, we provide a variant of the invariant hexagon argument from [7] that ensures robust forward invariance of suitable hexagons for the adaptive dynamics, under maximum bounds on the perturbation δ . This allows us to use the barrier Lyapunov function and strictification approach from [9] to get the stabilization, parameter identification, and integral ISS, which would not be possible using earlier methods.

IV. REVIEW OF NONADAPTIVE CASE

We next review the controller design from [17] and the ISS and strict Lyapunov function results from [7]. The controllers in [7], [17] have the form

$$u_2 = \frac{\kappa \cos(\phi)}{1 + \kappa \rho} - h'(\rho) \cos(\phi) + \mu \sin(\phi) \quad (2)$$

where $\mu > 0$ is a steering constant. In [7], h is any function that satisfies the following two assumptions:

Assumption 1: The function $h : (0, +\infty) \rightarrow [0, +\infty)$ is C^2 , h' has only finitely many zeros, $\lim_{\rho \rightarrow 0^+} h(\rho) = \lim_{\rho \rightarrow +\infty} h(\rho) = +\infty$, and there is a constant $\rho_0 > 0$ such that $h(\rho_0) = 0$. \square

Assumption 2: The function h satisfies Assumption 1 and:

- (i) There is an increasing C^1 function $\gamma : [0, +\infty) \rightarrow [\mu, +\infty)$ such that $\gamma(h(\rho)) \geq \max\{\mu, 1 + 0.5\mu^2 + h''(\rho)\}$ for all $\rho > 0$.
- (ii) There is a function $\Gamma \in \mathcal{K}_{\infty} \cap C^1$ such that $\Gamma(h(\rho)) \geq [h'(\rho)]^2$ for all $\rho > 0$.
- (iii) The function $h'(\rho)(\rho - \rho_0)$ is positive for all $\rho > 0$ with $\rho \neq \rho_0$.

Here $\rho_0 > 0$ is from Assumption 1. \square

Remark 1: The arguments in [7] show that Assumptions 1-2 hold for

$$h(\rho) = \alpha \left\{ \rho + \frac{\rho_0^2}{\rho} - 2\rho_0 \right\} \quad (3)$$

from [17] for any constants $\alpha > 0$ and $\rho_0 > 0$, if we take $\gamma(q) = 2(q + 2\alpha\rho_0)^3 / \{\alpha^2\rho_0^4\} + 1 + 0.5\mu^2 + \mu$ and $\Gamma(q) = \frac{18\alpha}{\rho_0} q + (\frac{2}{\rho_0})^4 (\frac{\rho_0}{\alpha^2}) q^4$. \square

The works [7], [17] assume that $K_2 = 1$, in which case substituting (2) into (1) produces the closed loop system

$$\begin{cases} \dot{\rho} &= -\sin(\phi) \\ \dot{\phi} &= h'(\rho) \cos(\phi) - \mu \sin(\phi) \end{cases} \quad (4)$$

on $\mathcal{X} = (0, +\infty) \times (-\pi/2, \pi/2)$ when no perturbation δ is present. Under Assumption 1, [17] shows GAS of (4) with respect to $((\rho_0, 0), \mathcal{X})$, using the nonstrict Lyapunov function

$$V(\rho, \phi) = -\ln(\cos(\phi)) + h(\rho) \quad (5)$$

for (4) with respect to $((\rho_0, 0), \mathcal{X})$ and LaSalle Invariance. For robustness, the arguments from [7] show that (4) admits the strict Lyapunov function

$$U_2(\rho, \phi) = -h'(\rho)\sin(\phi) + \frac{1}{\mu} \int_0^{V(\rho, \phi)} \gamma(m) dm + \Gamma(V(\rho, \phi)) + V(\rho, \phi) \quad (6)$$

with respect to $((\rho_0, 0), \mathcal{X})$ when V is given by (5). In fact, along all trajectories of (4) in \mathcal{X} , we have $U_2 \geq V$ and

$$\dot{U}_2 \leq -0.5[h'(\rho)\cos(\phi)]^2 - \sin^2(\phi) - [\Gamma'(V(\rho, \phi)) + 1]\mu \frac{\sin^2(\phi)}{\cos(\phi)}, \quad (7)$$

whose right side is negative definite with respect to $(\rho_0, 0)$. This allowed us to prove ISS results with respect to additive uncertainty on the controller, under certain restrictions on the magnitude of δ that maintain forward invariance properties.

V. ADAPTIVE CONTROLLER DESIGN

We aim to extend [7] to the constant speed adaptive curve tracking dynamics (1) where K_2 is to be identified (but see Remark 3 below for a further extension involving adaptive speed control with unknown control gains). In practice, positive lower and upper bounds for K_2 are known. Therefore, we assume the following in the rest of this paper:

Assumption 3: There are known constants $c_{\min} > 0$ and $c_{\max} > 0$ such that $c_{\min} < K_2 < c_{\max}$. \square

Following [9], we take the update law

$$\dot{\hat{K}}_2 = (\hat{K}_2 - c_{\min})(c_{\max} - \hat{K}_2)\bar{U}, \quad (8)$$

where the estimator \hat{K}_2 for the parameter K_2 has the state space (c_{\min}, c_{\max}) and the locally Lipschitz function \bar{U} is to be specified.¹ Denoting the adaptive controller by v gives

$$\begin{cases} \dot{\rho} &= -\sin(\phi) \\ \dot{\phi} &= \frac{\kappa \cos(\phi)}{1 + \kappa \rho} - K_2 v - K_2 \delta \\ \dot{\hat{K}}_2 &= (\hat{K}_2 - c_{\min})(c_{\max} - \hat{K}_2)\bar{U} \end{cases} \quad (9)$$

with state variables (ρ, ϕ, \hat{K}_2) . Choosing $v = u_2/\hat{K}_2$ and $\bar{U} = -(\partial U_2/\partial \phi)v$, where u_2 is the controller from (2) and U_2 is

¹A simple uniqueness of solutions argument shows that each trajectory of (8) starting in (c_{\min}, c_{\max}) remains in (c_{\min}, c_{\max}) at all positive times. To see why, suppose that $K_*(t)$ were a solution of (8) starting in (c_{\min}, c_{\max}) that reached c_{\max} at some positive time \bar{t} . Then the time reversed system would admit *two* distinct solutions for the initial state c_{\max} on $[0, \bar{t}]$, namely, the constant solution at c_{\max} and $t \mapsto K_*(\bar{t} - t)$, contradicting the uniqueness property for solutions. Similar reasoning applies to the other endpoint c_{\min} .

the strict Lyapunov function from (6), gives the augmented tracking and parameter estimation dynamics

$$\begin{cases} \dot{\rho} &= -\sin(\phi) \\ \dot{\phi} &= \left(1 - \frac{K_2}{\hat{K}_2}\right) \frac{\kappa \cos(\phi)}{1 + \kappa \rho} \\ &+ \frac{K_2}{\hat{K}_2} [h'(\rho)\cos(\phi) - \mu \sin(\phi)] - K_2 \delta \\ \dot{\hat{K}}_2 &= -(\hat{K}_2 - c_{\min})(c_{\max} - \hat{K}_2) \frac{\partial U_2}{\partial \phi} \frac{u_2}{\hat{K}_2}. \end{cases} \quad (10)$$

We wish to establish GAS and iISS properties for (10). However, the desired state space $\mathcal{X}^b = (0, +\infty) \times (-\pi/2, \pi/2) \times (c_{\min}, c_{\max})$ will not be forward invariant for (10) unless we impose restrictions on the magnitude of δ . To see why, choose the initial states $(\rho(0), \phi(0)) = (2\rho_0, 0)$, any initial value for \hat{K}_2 in $[c_{\min}, c_{\max}]$, and the constant perturbation

$$\delta = -\frac{1}{c_{\min}} \left[\frac{c_{\max}}{c_{\min}} (\kappa + \mu) + \frac{\pi}{2\rho_0} \right], \quad (11)$$

and suppose that $(\rho(t), \phi(t), \hat{K}_2(t))$ were the corresponding trajectory for (10) defined on $[0, \rho_0]$. Then $\rho(t) \geq \rho_0$ for all $t \in [0, \rho_0]$, so $h'(\rho(t)) \geq 0$ for all $t \in [0, \rho_0]$. Hence, $\dot{\phi}(t) \geq -\frac{c_{\max}}{c_{\min}} (\kappa + \mu) - K_2 \delta \geq \frac{\pi}{2\rho_0}$ for all $t \in [0, \rho_0]$, which would give $\phi(\rho_0) \geq \pi/2$. We turn to our restrictions on δ next.

VI. FORWARDLY INVARIANT HEXAGONS

We extend the forwardly invariant hexagon arguments from [7] to the adaptive dynamics (10), which will lead to iISS results for (10) under suitable restrictions on the magnitude of δ . Since there is no disturbance in the \hat{K}_2 dynamics in (10), maintaining forward invariance of \mathcal{X}^b only involves the (ρ, ϕ) subdynamics

$$\begin{cases} \dot{\rho} &= -\sin(\phi) \\ \dot{\phi} &= \cos(\phi) \left[\frac{\kappa}{1 + \kappa \rho} \left(1 - \frac{K_2}{\hat{K}_2}\right) + \frac{K_2}{\hat{K}_2} h'(\rho) \right] \\ &- \frac{K_2}{\hat{K}_2} \mu \sin(\phi) - K_2 \delta \end{cases} \quad (12)$$

of (10), where K_2 and its estimator \hat{K}_2 both lie on (c_{\min}, c_{\max}) as before. We now add the assumption that the function h satisfying Assumptions 1-2 also satisfies:

Assumption 4: The function h satisfies $\lim_{\rho \rightarrow 0^+} h'(\rho) = -\infty$ and $\liminf_{\rho \rightarrow \infty} h'(\rho) > 0$. Also, $h''(\rho_0) > 0$. \square

Notice that Assumption 4 holds for the choice (3) of h for any choices of its constants $\alpha > 0$ and $\rho_0 > 0$. Using Assumption 4, we can choose a sufficiently small constant $\rho_* > 0$ and a sufficiently large constant $\bar{K} > 0$ such that

$$\rho_* \in \left(0, \frac{\rho_0}{1 + c_{\max}/c_{\min}}\right) \text{ and } \bar{K} > 1 + \frac{c_{\max}}{c_{\min}} \quad (13)$$

hold, and such that

$$\min\{|h'(\rho)| : \rho_* \leq \rho \leq B_1\} > \kappa \left(\frac{c_{\max}}{c_{\min}}\right)^2 \quad (14)$$

and

$$\min\{h'(\rho) : \bar{K}\rho_0 \leq \rho \leq D_1\} > \frac{\kappa}{1 + \kappa \bar{K} \rho_0} \left(\frac{c_{\max}}{c_{\min}}\right)^2, \quad (15)$$

where $B_1 = \rho_*(1 + c_{\max}/c_{\min})$ and $D_1 = \bar{K}\rho_0 + \rho_* c_{\max}/c_{\min}$. (To see why we can satisfy (13)-(15), let I_0 denote the \liminf from Assumption 4. The left side of (14) goes to $+\infty$ as

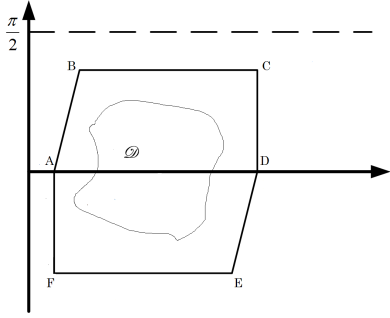


Fig. 1. Forwardly Invariant Hexagon $\tilde{H}(\rho_*, \mu, \bar{K})$ in (ρ, ϕ) Plane

$\rho_* \rightarrow 0^+$ so we can satisfy (14) for small enough $\rho_* > 0$, while the left side of (15) is above $I_0/2$ if \bar{K} is large enough. Then we can satisfy (15) by taking \bar{K} big enough to make the right side of (15) less than $I_0/2$.) Our condition on \bar{K} ensures that $\bar{K}\rho_0 > B_1$. Pick any constant $\mu \in (0, \pi/(2\rho_*))$ such that

$$\mu \tan(\mu\rho_*) > (c_{\max}/c_{\min})^2 \left[\kappa + |h'|_{[\rho_*, D_1]} \right], \quad (16)$$

by choosing μ close enough to but below $\pi/(2\rho_*)$.

Let $\tilde{H}(\rho_*, \mu, \bar{K})$ denote the region in the (ρ, ϕ) plane that is bounded by the hexagon having vertices $A = (\rho_*, 0)$, $B = (B_1, \mu\rho_*)$, $C = (D_1, \mu\rho_*)$, $D = (D_1, 0)$, $E = (\bar{K}\rho_0, -\mu\rho_*)$, and $F = (\rho_*, -\mu\rho_*)$. Its legs AB and ED have the slope $\mu^\sharp = c_{\min}\mu/c_{\max}$. These hexagons agree with the ones for the nonadaptive case in [7], except that the tilted legs in [7] have slope μ . For each compact subset \mathcal{D} of $\mathcal{X} = (0, +\infty) \times (-\pi/2, \pi/2)$, we can choose ρ_* , \bar{K} , and μ to satisfy (13)-(16) and to be such that $\mathcal{D} \subseteq \tilde{H}(\rho_*, \mu, \bar{K})$. This can be done by moving $\rho_* > 0$ far enough to the left and $\bar{K}\rho_0$ far enough to the right, and making μ large enough and close enough to $\pi/(2\rho_*)$ so that the hexagon approximates a rectangle whose top and bottom are as close as desired to $\pm\pi/2$. See Fig. 1. Given any compact subset $\mathcal{D} \subseteq \mathcal{X}$, our work will establish robust forward invariance of any hexagon that contains \mathcal{D} .

To simplify the analysis, we set

$$\begin{aligned} Q_1(\rho, \phi, \hat{K}_2, K_2) &= \\ \cos(\phi) \left(\frac{\kappa}{1+\kappa\rho} \frac{\hat{K}_2 - K_2}{\hat{K}_2} + \frac{K_2}{\hat{K}_2} h'(\rho) - \frac{K_2}{\hat{K}_2} \mu \tan(\phi) \right) \text{ and} \quad (17) \\ Q_2(\rho, \phi, \hat{K}_2, K_2) &= Q_1(\rho, \phi, \hat{K}_2, K_2) + \mu^\sharp \sin(\phi). \end{aligned}$$

We use the following lemma [8]:

Lemma 1: For all constants $K_2 \in [c_{\min}, c_{\max}]$ and $\hat{K}_2 \in [c_{\min}, c_{\max}]$, we have: (a) $Q_1(\rho, \phi, \hat{K}_2, K_2) < 0$ (resp., > 0) for all $(\rho, \phi) \in BC$ (resp., FE). (b) $Q_2(\rho, \phi, \hat{K}_2, K_2) < 0$ (resp., > 0) for all $(\rho, \phi) \in AB$ (resp., ED). \square

By Lemma 1, the following constants are positive:

$$\begin{aligned} \Delta_* &= \\ \min \{ |Q_1(\rho, \phi, \hat{K}_2, K_2)| / K_2 : \\ (\rho, \phi) \in BC \cup EF, K_2 \in [c_{\min}, c_{\max}], \hat{K}_2 \in [c_{\min}, c_{\max}] \} \quad (18) \end{aligned}$$

$$\begin{aligned} \Delta_{**} &= \\ \min \{ |Q_2(\rho, \phi, \hat{K}_2, K_2)| / K_2 : \\ (\rho, \phi) \in AB \cup ED, K_2 \in [c_{\min}, c_{\max}], \hat{K}_2 \in [c_{\min}, c_{\max}] \} \quad (19) \end{aligned}$$

Fix any constant $\delta_* \in (0, \min\{\Delta_*, \Delta_{**}\})$. We prove the following robust forward invariance result for (12):

Theorem 1: Let Assumptions 1-4 hold. Let ρ_* and \bar{K} be any constants satisfying (13)-(15). Choose any constant $\mu \in (0, \pi/(2\rho_*))$ satisfying (16), any constant $K_2 \in [c_{\min}, c_{\max}]$, any function $\hat{K}_2 : [0, \infty) \rightarrow [c_{\min}, c_{\max}]$, and any measurable function $\delta : [0, \infty) \rightarrow [-\delta_*, \delta_*]$. Then all trajectories of (12) starting in $\tilde{H}(\rho_*, \mu, \bar{K})$ stay in $\tilde{H}(\rho_*, \mu, \bar{K})$ for all $t \geq 0$. \square

Proof: We prove that for each such trajectory $(\rho, \phi)(t)$ of (12) starting in $\tilde{H}(\rho_*, \mu, \bar{K})$, the vector field $(\dot{\rho}, \dot{\phi})$ points into the hexagon along all six legs of the hexagon, so the trajectory cannot leave the hexagon. By part (a) of Lemma 1, we have $\dot{\phi} = Q_1 - K_2\delta \leq Q_1 + K_2\delta_* < 0$ on BC because $\delta_* < \Delta_*$. Along FE , part (a) of Lemma 1 gives $\dot{\phi} = Q_1 - K_2\delta \geq Q_1 - K_2\delta_* > 0$ along FE , again because $\delta_* < \Delta_*$. Hence, the trajectories point down (resp., up) along the top (resp., bottom) leg. Along AF , we have $\dot{\rho} > 0$, except at the point A . At A , we have $\dot{\rho} = 0$ and $\dot{\phi} < 0$, because $\delta_* < \Delta_{**}$, which makes ρ increase. Along the leg CD , we have $\dot{\rho} < 0$, except at D . At D , we have $\dot{\rho} = 0$ and $\dot{\phi} > 0$, because $\delta_* < \Delta_{**}$, which makes ρ decrease. Hence we cannot exit the hexagon through its top, bottom, or vertical legs.

To rule out exits through the tilted legs, we use $\mathcal{S}(\rho, \phi) = \phi - \mu^\sharp\rho$. For each pair $(\rho, \phi) \in \mathcal{X}$, the value $\mathcal{S}(\rho, \phi)$ is the ϕ -axis intercept of the line through (ρ, ϕ) having slope μ^\sharp , so the theorem will follow once we show that $\dot{\mathcal{S}} < 0$ (resp., $\dot{\mathcal{S}} > 0$) along AB (resp., ED). Since $\delta_* < \Delta_{**}$, it follows from part (b) of Lemma 1 that along AB , we have $\dot{\mathcal{S}} = Q_2 - K_2\delta < 0$. This rules out exits along AB . Again using part (b) of Lemma 1 and the fact that $\delta_* < \Delta_{**}$, we also have $\dot{\mathcal{S}} = Q_2 - K_2\delta > 0$ for all $(\rho, \phi) \in DE$. This rules out exits along DE , completing the proof. \blacksquare

Remark 2: Given any function h satisfying Assumptions 1, 2, and 4, the bound $\min\{\Delta_*, \Delta_{**}\}$ is the supremum of all possible perturbation bounds that render $\tilde{H}(\rho_*, \mu, \bar{K})$ robustly forward invariant for (12) for all K_2 and all signals $\hat{K}_2(t)$, and therefore is optimal. In fact, for each constant $\tilde{\delta} > \min\{\Delta_*, \Delta_{**}\}$, we can find a point \tilde{p} on the boundary of $\tilde{H}(\rho_*, \mu, \bar{K})$ such that the trajectory of (12) starting at \tilde{p} for one of the constant perturbations $\pm\tilde{\delta}$ and some admissible choices of K_2 and \hat{K}_2 exits the hexagon. To see why, note that if $\min\{\Delta_*, \Delta_{**}\} = \Delta_{**}$ and if $\tilde{\delta} > \Delta_{**} = -Q_2(\rho, \phi, \hat{K}_2, K_2)/K_2$ at some $\tilde{p} = (\rho, \phi)^\top \in AB$, then taking the constant disturbance $\delta = -\tilde{\delta}$ gives $\dot{\mathcal{S}} = Q_2(\rho, \phi, \hat{K}_2, K_2) + K_2\tilde{\delta} > 0$ at \tilde{p} , so we exit from \tilde{p} . On the other hand, if $\tilde{\delta} > \Delta_{**} = Q_2(\rho, \phi, \hat{K}_2, K_2)/K_2$ at some $\tilde{p} = (\rho, \phi)^\top \in ED$, then taking the constant disturbance $\delta = \tilde{\delta}$ gives $\dot{\mathcal{S}} = Q_2(\rho, \phi, \hat{K}_2, K_2) - K_2\tilde{\delta} < 0$, so we again exit at \tilde{p} . The other cases where $\min\{\Delta_*, \Delta_{**}\} = \Delta_*$ are handled similarly. Hence, for each choice of h , our bound δ_* is optimal for maintaining the necessary invariance properties of $\tilde{H}(\rho_*, \mu, \bar{K})$ for (12) and not at all conservative, and our controller is independent of the choice of the hexagon. \square

VII. INTEGRAL INPUT-TO-STATE STABILITY

The preceding section shows that for each hexagon $\tilde{H}(\rho_*, \mu, \bar{K})$ and each bound δ_* on the admissible actuator

error δ satisfying the requirements from Theorem 1, the augmented tracking and parameter estimation dynamics

$$\begin{cases} \dot{\tilde{q}}_1 &= -\sin(\tilde{q}_2) \\ \dot{\tilde{q}}_2 &= \frac{\kappa \cos(\tilde{q}_2)}{1+\kappa(\tilde{q}_1+\rho_0)} - \frac{K_2}{\tilde{K}_2} u_2 - K_2 \delta \\ \dot{\tilde{K}}_2 &= -(\tilde{K}_2 - c_{\min})(c_{\max} - \tilde{K}_2) \frac{\partial U_2}{\partial \phi} \frac{u_2}{\tilde{K}_2} \end{cases} \quad (20)$$

with state variables $\tilde{q} = (\tilde{q}_1, \tilde{q}_2) = (\rho - \rho_0, \phi)$ and \tilde{K}_2 and disturbances δ valued in $\mathcal{U} = [-\delta_*, \delta_*]$ has the robustly forward invariant set

$$\mathcal{X}^+ = (\tilde{H}(\rho_*, \mu, \bar{K}) - \mathcal{E}) \times (c_{\min}, c_{\max}). \quad (21)$$

Here u_2 is the controller from (2), U_2 is the strict Lyapunov function from (6), and we subtract the equilibrium $\mathcal{E} = (\rho_0, 0)$ in (21) to account for going from the original curve tracking dynamics for $q = (\rho, \phi)$ to the error dynamics for \tilde{q} . It is therefore meaningful to ask whether the dynamics

$$\begin{cases} \dot{\tilde{q}}_1 &= -\sin(\tilde{q}_2) \\ \dot{\tilde{q}}_2 &= \frac{\kappa \cos(\tilde{q}_2)}{1+\kappa(\tilde{q}_1+\rho_0)} - u_2 + \frac{\tilde{K}_2}{\tilde{K}_2 + K_2} u_2 - K_2 \delta \\ \dot{\tilde{K}}_2 &= -(\tilde{K}_2 + K_2 - c_{\min})(c_{\max} - \tilde{K}_2 - K_2) \frac{\partial U_2}{\partial \phi} \frac{u_2}{\tilde{K}_2 + K_2} \end{cases} \quad (22)$$

for $(\tilde{q}, \tilde{K}_2) = (\rho - \rho_0, \phi, \tilde{K}_2 - K_2)$ is iISS with respect to $(\mathcal{U}, 0, \mathcal{X}^\#)$ with the choices $\mathcal{U} = [-\delta_*, \delta_*]$ and

$$\mathcal{X}^\# = (\tilde{H}(\rho_*, \mu, \bar{K}) - \mathcal{E}) \times (c_{\min} - K_2, c_{\max} - K_2), \quad (23)$$

since this would give stabilization of the equilibrium point $(\rho_0, 0)$ and parameter identification when $\delta = 0$, as well as robustness in the sense of iISS on each hexagon. This is much stronger than local stabilization, because the hexagons can be arbitrarily wide. In [8], we use ideas from [9] to prove that the iISS property is indeed satisfied. Here we summarize the ideas behind our iISS result. Throughout this section, we fix a hexagon $\tilde{H}(\rho_*, \mu, \bar{K})$ and a corresponding disturbance bound δ_* as in Theorem 1, so all disturbances δ are valued in $\mathcal{U} = [-\delta_*, \delta_*]$. To simplify the notation, we sometimes denote the hexagon simply by \tilde{H} .

Following [9], we first take the barrier nonstrict Lyapunov function $U_3 : \mathcal{X}^\# \rightarrow [0, +\infty)$ defined by

$$\begin{aligned} U_3(\tilde{q}, \tilde{K}_2) &= \\ U_2(\tilde{q} + (\rho_0, 0)) &+ \int_0^{\tilde{K}_2} \frac{\ell}{(\ell + K_2 - c_{\min})(c_{\max} - \ell - K_2)} d\ell. \end{aligned} \quad (24)$$

A partial fraction decomposition shows that the integral in (24) can be rewritten as

$$\frac{c_{\min} - K_2}{c_{\max} - c_{\min}} \ln \left| \frac{\tilde{K}_2 + K_2 - c_{\min}}{\tilde{K}_2 - c_{\min}} \right| - \frac{c_{\max} - K_2}{c_{\max} - c_{\min}} \ln \left| \frac{c_{\max} - \tilde{K}_2 - K_2}{c_{\max} - K_2} \right|$$

and so is a modulus with respect to $(0, \mathcal{W})$ where $\mathcal{W} = (c_{\min} - K_2, c_{\max} - K_2)$. Moreover, the decay estimate (7) along all trajectories of (4) gives

$$\begin{aligned} \dot{U}_3 &\leq -0.5 [h'(\rho) \cos(\phi)]^2 - \sin^2(\phi) \\ &- [\Gamma'(V(\rho, \phi)) + 1] \mu \frac{\sin^2(\phi)}{\cos(\phi)} - K_2 \frac{\partial U_2}{\partial \phi}(\rho, \phi) \delta \end{aligned} \quad (25)$$

along all trajectories of (22). Since $K_2(\partial U_2/\partial \phi)$ is bounded on $\mathcal{X}^\#$, condition (25) resembles the iISS Lyapunov function

decay condition [1], save for the fact that its right side is not negative definite in the full state (\tilde{q}, \tilde{K}_2) when $\delta = 0$. Also, the adaptive arguments from [9] do not apply directly because of the restricted state space and the actuator error. On the other hand, we can use U_3 to prove [8]:

Theorem 2: Let h, ρ_*, μ, \bar{K} , and δ_* satisfy Assumptions 1-2 and the requirements of Section VI. Then (22) is iISS with respect to $([-\delta_*, \delta_*], 0, \mathcal{X}^\#)$. \square

Remark 3: The preceding analysis assumed that the speed of the robot is constant. However, we can extend it to include adaptive speed control. For details, see [8]. \square

VIII. SIMULATIONS

To illustrate our stabilization, parameter identification, and iISS results, we simulated (22) with the choice

$$h(\rho) = 10 \left(\rho + \frac{1}{\rho} - 2 \right). \quad (26)$$

This satisfies our assumptions with $\alpha = 10$ and $\rho_0 = 1$; see Remark 1. We used the functions γ and Γ from Remark 1 to generate the adaptive controls. We also took $c_{\min} = 0.5$, $c_{\max} = 0.75$, $\kappa = 2$, $\rho_* = 0.12$, $\mu = 0.995\pi/(2\rho_*) = 13.025$ and $\bar{K} = 10$, which give the hexagon $\tilde{H}(0.12, 13.025, 10) \subseteq \mathcal{X}$. These parameter values satisfy our requirements (13)-(16). Using the notation from above, our assumptions are satisfied with $\delta_* = 0.103671$. In our first simulation, we took the initial values $(\tilde{q}, \tilde{K}_2, \tilde{K}_2)(0) = ((0.2, 0.4), 0.2, 0.7)$ and the disturbance $\delta \equiv 0$ and obtained the trajectory components for $(\tilde{q}, \tilde{K}_2)(t)$ in Figure 2, which shows rapid convergence of the state and parameter estimation errors to zero. Our second simulation was the same as our first, except we included the sinusoidal disturbance $\delta(t) = 0.1 \sin(t)$. In Figure 3, we show the simulated trajectory for $\tilde{q}(t)$ for the second simulation. We do not show the \tilde{K}_2 trajectory for our second simulation with the disturbance, because it was the same as for the undisturbed case. With the nonzero disturbance, the errors $\tilde{q}(t)$ no longer converge to zero. Instead, there is a small overshoot in $\tilde{q}(t)$, which is made precise by the corresponding overshoot in the iISS estimate. This illustrates the effects of introducing a disturbance in the adaptive stabilization and parameter estimation dynamics.

IX. CONCLUSIONS

Adaptive stabilization and parameter identification for robotic curve tracking models with unknown control gains under controller uncertainty is a challenging problem that is beyond the scope of the standard adaptive control methods. This is due to the restricted state space, the need for strict Lyapunov functions for the augmented stabilization and estimation dynamics, and the need to identify the parameters instead of merely ensuring that the parameter estimate stays bounded. We used recently developed barrier Lyapunov function and strictification approaches to solve the stabilization and parameter identification problems and prove robustness under controller uncertainty in the sense of integral input-to-state stability. We aim to extend our work to curve tracking problems under time delays, and to three-dimensional curve tracking.

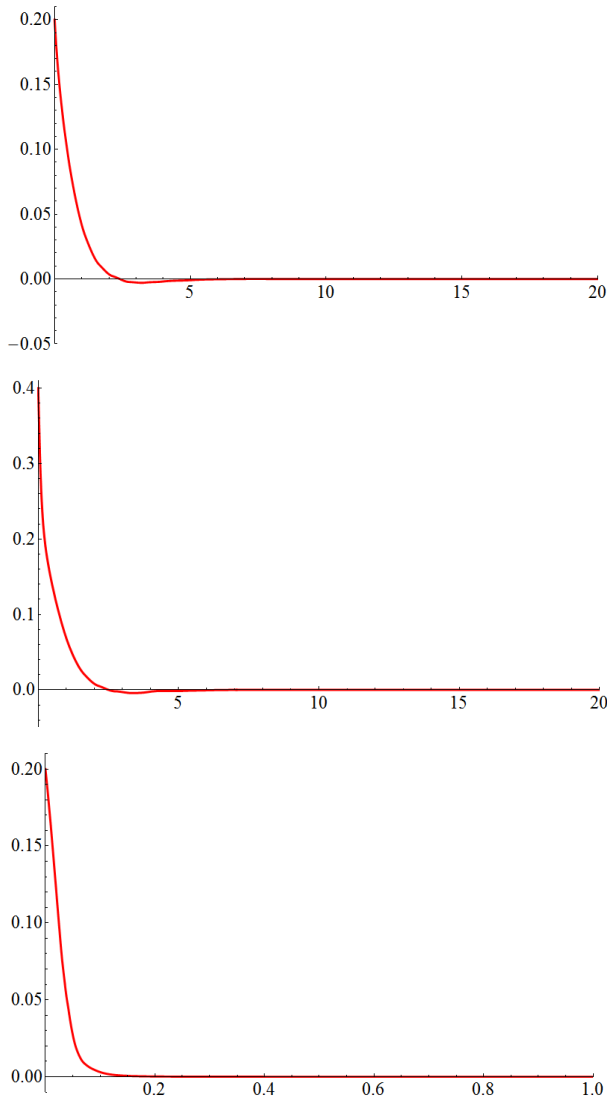


Fig. 2. Simulated Trajectory Components $\rho(t) - \rho_0$ (Top), $\phi(t)$ (Middle), and \bar{K}_2 (Bottom) with Disturbance $\delta \equiv 0$.

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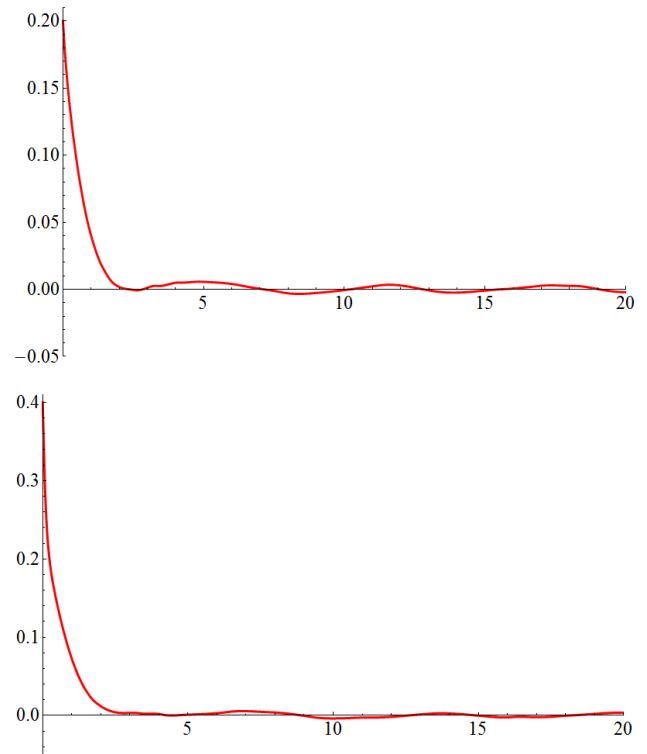


Fig. 3. Simulated Trajectory Components $\rho(t) - \rho_0$ (Top) and $\phi(t)$ (Bottom) with Disturbance $\delta = 0.1 \sin(t)$.

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